

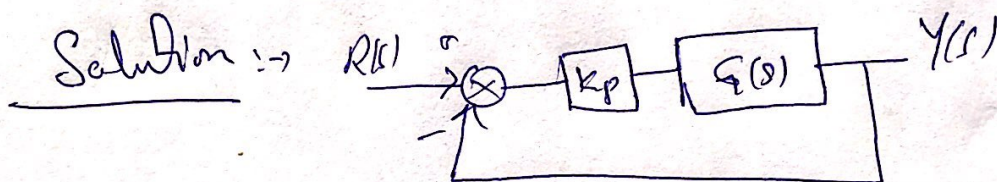
Time Response (Example)

①

Ex: The plant t/f $G(s)$ is given as

$$G(s) = \frac{1}{s(s+4)}$$

If this plant is kept in a unity feedback configuration with a proportional controller $K_p = 16$, then find t_r , t_s and M_p ?



$$H(s) = \frac{Y(s)}{R(s)} = \frac{K_p G(s)}{1 + K_p G(s)} = \frac{16 \frac{1}{s(s+4)}}{1 + \frac{16}{s(s+4)}}$$
$$= \frac{16}{s^2 + 4s + 16} \rightarrow \textcircled{1}$$

The std. form is

$$H(s) = \frac{\omega_n^2}{s^2 + 2\zeta\omega_n s + \omega_n^2} \rightarrow \textcircled{2}$$

Compare ① & ②

$$\omega_n^2 = 16 \Rightarrow \omega_n = 4$$
$$2\zeta\omega_n = 4 \Rightarrow \zeta\omega_n = 2 \Rightarrow \zeta = \frac{2}{4} = 0.5$$

$$\zeta = 0.5 \quad \omega_n = 4$$

(2)

$$t_r \approx \frac{1.8}{\omega_n} = \frac{1.8}{4} = 0.45 \text{ Sec}$$

$$t_s = \frac{4.6}{\zeta \omega_n} = \frac{4.6}{0.5 \times 4} = 2.3 \text{ Sec}$$

$$M_p = 100 \times e^{-\pi \zeta / \sqrt{1-\zeta^2}} \%$$

$$= 100 \times e^{-\pi \frac{0.5}{\sqrt{1-0.5^2}}} = ? \%$$

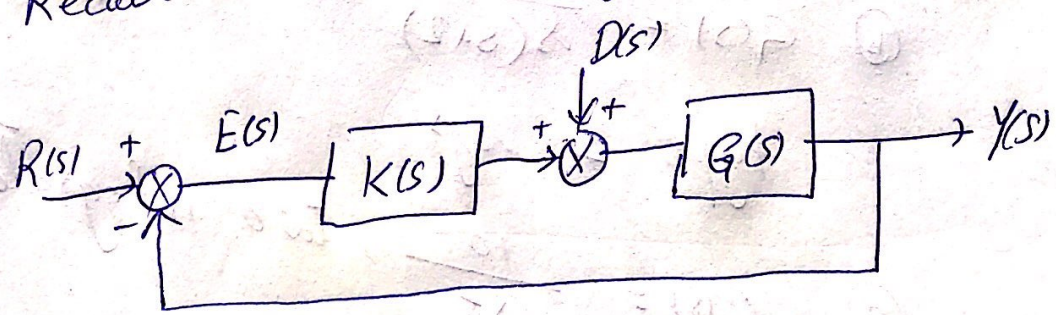
→ The specifications we've studied so far (i.e., t_r , t_s and m_p) are transient response specifications.

→ In tracking references we also keep in mind the steady state steady state! → "when all the transients die out and the sys. comes to rest or uniform motion".

⇒ "We need (ideally) zero error in steady state".

Steady State Error

Recall the block diagram



→ A control sys. is required s.t $E(s) \rightarrow 0$ i.e., $Y(s) = R(s)$

$$\frac{E(s)}{R(s)} = \frac{1}{1 + G(s)K(s)}$$

~~$$\frac{E(s)}{D(s)} = \frac{G(s)}{1 + G(s)K(s)}$$~~

⇒ We want $E(s)$ to be 0 ~~even~~ in the presence of disturbance.

System Type ⇒ Number of integrators in a system defines the type of sys. e.g.,

(a) $G(s) = \frac{1}{s+1}$, no integrator
type 0 sys.

(b) $G(s) = \frac{1}{s(s+2)}$, One integrator
type 1 sys.

(c) $G(s) = \frac{10}{s^2}$, two integrators
type 2 sys.

Steady state error (E_{ss})

(5)

$$\frac{E(s)}{R(s)} = \frac{1}{1 + G(s)K(s)}$$

$$E(s) = \frac{1}{1 + G(s)K(s)} R(s)$$

By FVT

$$E_{ss} = \lim_{s \rightarrow 0} s E(s)$$

① $E_{ss} = ?$ when $R(s)$ is ^{unit} step i.e., $R(s) = \frac{1}{s}$

$$E_{ss} = \lim_{s \rightarrow 0} s \cdot \frac{1}{s} \cdot \frac{1}{1 + G(s)K(s)}$$

$$E_{ss} = \frac{1}{1 + G(0)K(0)} \rightarrow \text{①}$$

If $G(s)K(s)$ is type 0 then

$E_{ss} \neq 0$ For $E_{ss} = 0$ we need

$G(0)K(0) = \infty$ which is possible only

if $G(s)K(s)$ has an integrator
i.e., an integrator is required either
in the plant ~~or~~ $G(s)$ or in the
controller $K(s)$.

Now E_{ss} due to $D(s)$ [$D(s) = \frac{1}{s}$] (6)

$$E_{ss} = - \lim_{s \rightarrow 0} s \cdot \frac{1}{s} \cdot \frac{G(0)}{1 + G(s)K(s)}$$

$$= - \lim_{s \rightarrow 0} \frac{1}{\frac{1}{G(s)} + K(s)}$$

$$= - \frac{1}{\frac{1}{G(0)} + K(0)}$$

→ Now an integrator in the plant is not enough.

→ In order ~~that~~ for $E_{ss} = 0$ we must have an integrator in the controller $K(s)$.

(2) E_{ss} in the presence of Ramp ($\frac{1}{s^2}$) i/p

$$E(s) = \frac{1}{s^2} \frac{1}{1 + G(s)K(s)}$$

$$E_{ss} = \lim_{s \rightarrow 0} s \cdot \frac{1}{s^2} \frac{1}{1 + G(s)K(s)}$$

$$= \lim_{s \rightarrow 0} \frac{1}{s + sG(s)K(s)}$$

(a) if $G(s)K(s)$ is type 0 then $E_{ss} = \infty$

(b) if $G(s)K(s)$ is type 1 then $sG(0)K(0)$ may be some finite value (There will be a finite error in this case)

(c) If $G(s)K(s)$ is type 2

then $\lim_{s \rightarrow 0} s G(s)K(s) = \infty$

$\Rightarrow E_{ss} = 0$

Ramp disturbance $D(s) = \frac{1}{s^2}$

$$\frac{E(s)}{D(s)} = \frac{-G(s)}{1 + G(s)K(s)}$$

$$E(s) = -D(s) \frac{G(s)}{1 + G(s)K(s)}$$

$$E_{ss} = -\lim_{s \rightarrow 0} s \frac{1}{s^2} \frac{G(s)}{1 + G(s)K(s)} \quad \because D(s) = \frac{1}{s^2}$$

$$= -\lim_{s \rightarrow 0} \frac{G(s)}{s + sG(s)K(s)}$$

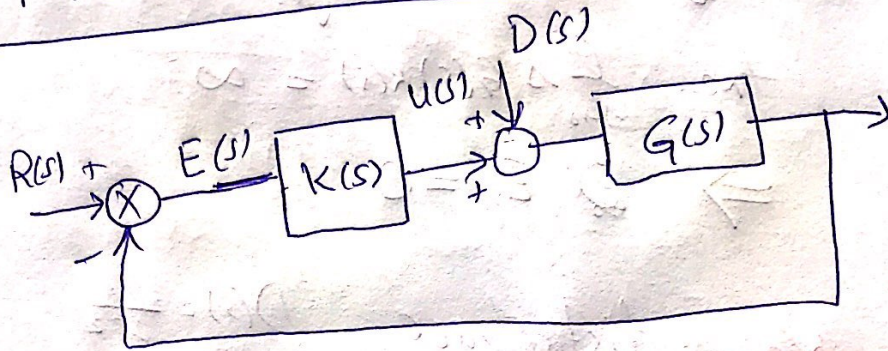
$$= -\lim_{s \rightarrow 0} \frac{1}{\frac{s}{G(s)} + sK(s)}$$

Now for complete disturbance

rejection we at two integrators in the controller $K(s)$.

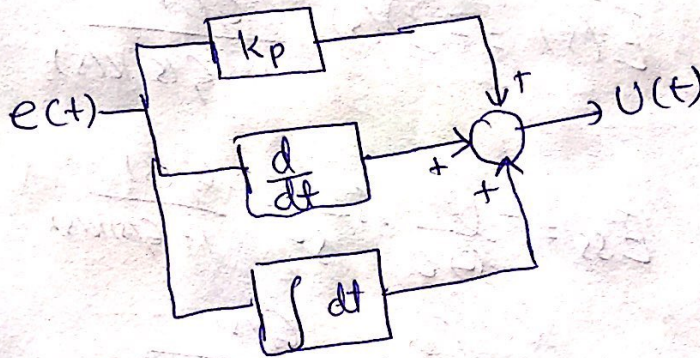
The same procedure can be carried out for parabolic i/p's.

PID Controller



$$U(t) = k_p e(t) + k_d \dot{e}(t) + k_i \int e(t) dt \quad \text{--- (1)}$$

↳ PID control in Time-domain.



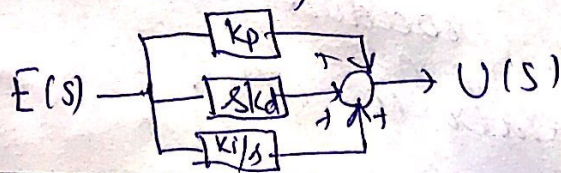
Taking Laplace of (1)

$$U(s) = k_p E(s) + k_d s E(s) + \frac{k_i}{s} E(s)$$

$$= \left[k_p + k_d s + \frac{k_i}{s} \right] E(s)$$

$$\frac{U(s)}{E(s)} = k_p + k_d s + \frac{k_i}{s}$$

↳ TF of PID.



(9)

→ Now each component of PID has a role to play.

e.g. (a) $\int e(t) dt$ The integral term

$\int e(t) dt \neq 0$ and keep increasing

until $e(t) = 0$ (that's ~~what~~

why integral component is

imp. for steady state error

elimination (good tracking)

(b) The derivative term $\frac{d e(t)}{dt}$

→ As we know that $\frac{d e(t)}{dt}$ will be bigger if $e(t)$ changes more.

→ So derivative term is required to add damping to the sys.

(c) Proportional component $e(t)$

$K_p e(t) \uparrow$ if $|e(t)| \uparrow$

i.e., in order to make the error vanish quickly ~~we~~ (to make sys fast) we need bigger K_p .

(10)

$K_p \uparrow \longrightarrow$ Sys. is fast

$K_d \uparrow \longrightarrow$ Sys dampout quickly

$K_i \uparrow \longrightarrow$ Low steady state error.

Example \Rightarrow The plant t/f is given as

$$G(s) = \frac{1}{s(s+2)}$$

If $K(s) = K_p$ then

① Find E_{ss} to step Reference?

② " " " " Disturbance?

Solution \Rightarrow ①

$$\frac{E(s)}{R(s)} = \frac{1}{1 + G(s)K(s)}$$

$$E(s) = R(s) \frac{1}{1 + G(s)K(s)}$$

$$= \frac{1}{s} \frac{1}{1 + G(s)K(s)}$$

$$E_{ss} = \lim_{s \rightarrow 0} s E(s) = \lim_{s \rightarrow 0} s \frac{1}{s} \frac{1}{1 + G(s)K_p}$$

(1)

$$E_{ss} = \frac{1}{1 + G(0)K_p}$$

$$G(0) = \frac{1}{0(0+2)} = \infty$$

So

$$E_{ss} = \frac{1}{1 + \infty} = 0$$

$E_{ss} = 0$ in tracking a reference step

(2)

$$\frac{E(s)}{D(s)} = \frac{-G(s)}{1 + G(s)K_p}$$

$$E_{ss} = -\lim_{s \rightarrow 0} s \frac{1}{s} \frac{G(s)}{1 + G(s)K_p}$$

$$= -\lim_{s \rightarrow 0} \frac{G(s)}{1 + G(s)K_p}$$

$$= -\lim_{s \rightarrow 0} \frac{1}{\frac{1}{G(s)} + K_p}$$

$$= -\frac{1}{\frac{1}{G(0)} + K_p} = -\frac{1}{0 + K_p} \quad \because G(0) = \infty$$

$$E_{ss} = -\frac{1}{K_p} \neq 0$$

⇒ Conclusion: → Integral control is required to reject disturbance.

(12)

Example \Rightarrow For the given plant

$$G(s) = \frac{2}{s(s+2)}$$

Design a controller s.t

(a) $t_s = 4$

(b) $M_p = 10\%$

Solution

$$\frac{E(s)}{R(s)} = \frac{G(s)K(s)}{1 + G(s)K(s)}$$

$$= \frac{2K(s)}{s^2 + 2s + K(s)} \rightarrow \textcircled{1}$$

In $\textcircled{1}$ if $K(s) = K_p$ then we can only alter the rise time so

let $K(s) = K_p + K_d s$

then

$$\frac{E(s)}{R(s)} = \frac{2(K_p + K_d s)}{s^2 + 2s + K_p + K_d s}$$

$$= \frac{2(K_p + K_d s)}{s^2 + (2 + K_d)s + K_p} \rightarrow \textcircled{2}$$

Now we'll compare the denominator of (2) with the standard 2nd order

Hf

$$H(s) = \frac{\omega_n^2}{s^2 + 2\xi\omega_n s + \omega_n^2} \rightarrow (3)$$

$$\omega_n^2 = k_p \rightarrow (4)$$

$$k_d = 2\xi\omega_n - 2 \rightarrow (5)$$

$$2 + k_d = 2\xi\omega_n \Rightarrow \xi = \frac{2 + k_d}{2\omega_n}$$

Notice that though the numerators of (2) and (3) are not alike

and the formulas

$$t_r \approx \frac{1.8}{\omega_n} \quad \text{and} \quad M_p = e^{-\frac{\pi\xi}{\sqrt{1-\xi^2}}}$$

are valid for (3) only. Yet

using these formulas we can

get a very good starting

point for further iterations.

(14)

→ From give specs.

$$\frac{1.8}{\omega_n} = 4 \Rightarrow \omega_n = \frac{1.8}{4}$$

put this in (4)

$$\left(\frac{1.8}{4}\right)^2 = K_p \quad \text{and similarly}$$

We can workout ξ for ξ .

$$K_d = 2\xi\omega_n - 2$$

$$= 2\xi\left(\frac{1.8}{4}\right) - 2$$

ξ will come from

$$0.1 = e^{-\pi\xi/\sqrt{1-\xi^2}}$$

$$\ln(0.1) = \frac{-\pi\xi}{\sqrt{1-\xi^2}}$$

$$\left(\sqrt{1-\xi^2}\right) \ln(0.1) = -\pi\xi$$

$$(1-\xi^2) (\ln(0.1))^2 = \pi^2 \xi^2$$

$$(\ln(0.1))^2 - \xi^2 (\ln(0.1))^2 - \pi^2 \xi^2 = 0$$

$$(\ln(0.1))^2 = \xi^2 \left[(\ln(0.1))^2 - \pi^2 \right]$$

$$\xi = \sqrt{\frac{(\ln(0.1))^2}{\ln(0.1)^2 - \pi^2}} \quad \text{approx.}$$

Example (Satellite attitude control)

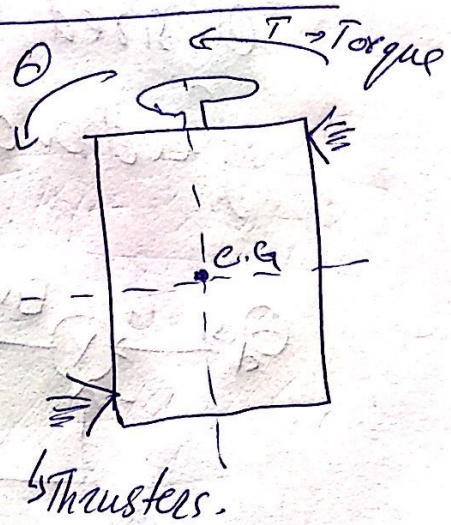
① Design a controller

s.t

$$t_r \leq 2 \text{ Sec} ?$$

$$M_p \leq 5\%$$

② $E_{ss} = 0$ in presence of the step disturbance?



Solution

EOM:

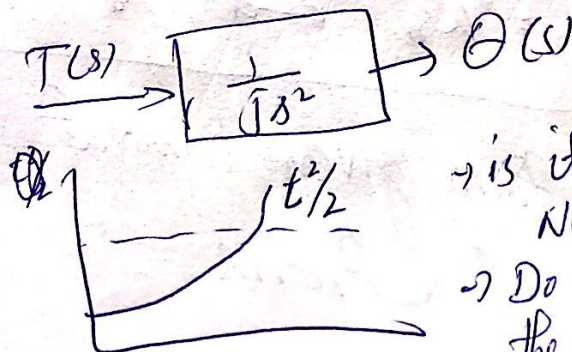
$$T(t) = J \ddot{\theta}$$

↳ moment of inertia.

$$\frac{\Theta(s)}{T(s)} = \frac{1}{Js^2}$$

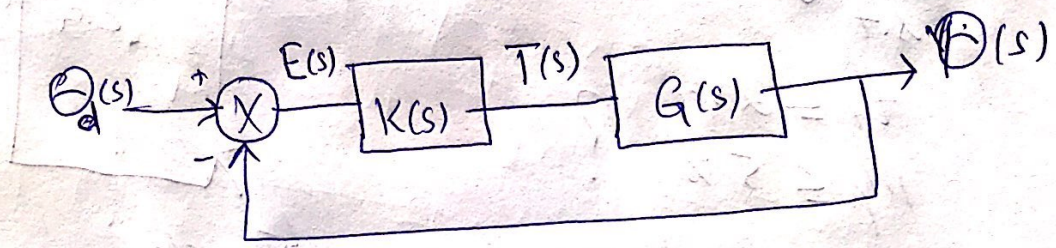
↳ Double integrator plant

In open loop



→ is it stable?
 NO
 → Do you like the response?

⇒ If we want to track a desired angle we'll need feedback.



$$\frac{\Theta(s)}{\Theta_d(s)} = \frac{K(s)G(s)}{1 + K(s)G(s)}$$

and let $K(s) = K_p + K_d s$
 i.e., PD control.

then (For simplicity assume $J=1$)

$$\frac{\Theta(s)}{\Theta_d(s)} = \frac{K_p + K_d s}{s^2 + K_d s + K_p} \rightarrow (1)$$

Comparing the denominator of (1) with the std. one $(s^2 + 2\zeta\omega_n s + \omega_n^2)$

$$K_p = \omega_n^2 \text{ and } K_d = 2\zeta\omega_n \rightarrow (2)$$

Now from given transient response specifications

~~1.8~~

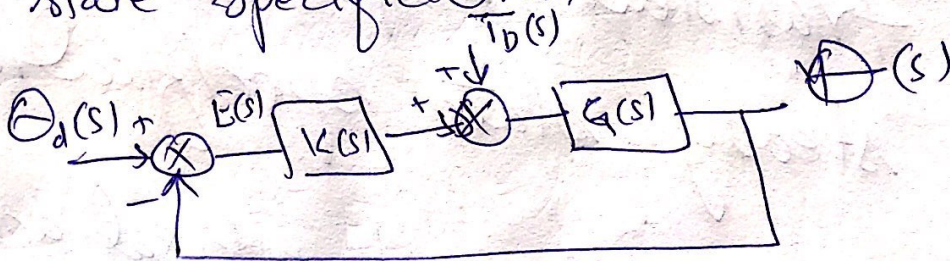
$$t_r \approx \frac{1.8}{\omega_n} \leq 2 \Rightarrow \omega_n \geq \frac{1.8}{2}$$

$$\omega_n \geq 0.9$$

$$\rightarrow m_p \leq 5\% \Rightarrow \xi \geq 0.707$$

Put this m_p & ξ in (2) will give values of k_p and K_p .

However, we also have steady state specification.



~~$$E(s) = \frac{D_d(s)}{1 + G(s)K(s)} - \frac{T_D(s)G(s)}{1 + G(s)K(s)}$$~~

$$D_d(s) = \frac{1}{s}, \quad T_D(s) = \frac{1}{s}$$

~~$$E_{ss} = \lim_{s \rightarrow 0} \left[s \frac{1}{s} \frac{1}{1 + G(s)K(s)} - s \frac{1}{s} \frac{G(s)}{1 + G(s)K(s)} \right]$$~~

$$= \frac{1}{1 + G(0)K(0)} - \lim_{s \rightarrow 0} \frac{1}{\frac{1}{G(s)} + K(s)}$$

$$= 0 - \frac{1}{0 + K_p}$$

$$\hookrightarrow K(0) = K(0) + K_p$$

$$\hookrightarrow G(s) = \frac{1}{s^2}$$

$$G(0) = \infty$$

Now we can increase K_p to reduce E_{ss} however,

increasing K_p will destroy

the transient response

⇒ PD control can't alone

cater for both the transient response and

steady state specifications.

⇒ So we'll need integral action

$$K(s) = K_p + K_d s + \frac{K_i}{s}$$

$$1 + G(s)K(s) = 1 + \frac{1}{s^2} \left(\frac{K_p s + K_d s^2 + K_i}{s} \right)$$

$$= \frac{s^3 + K_d s^2 + K_p s + K_i}{s^3}$$

Now

$$\begin{aligned} \frac{-E(s)}{T_D(s)} &= \frac{-G(s)}{1 + G(s)K(s)} = \frac{-\frac{1}{s^2}}{\frac{s^3 + K_d s^2 + K_p s + K_i}{s^3}} \\ &= \frac{-s}{s^3 + K_d s^2 + K_p s + K_i} \end{aligned}$$

E_{ss} for step disturbance $T_d(s) = \frac{1}{s}$

$$E_{ss} = \lim_{s \rightarrow 0} \frac{1}{s} \frac{1}{s} \frac{s}{s^3 + K_d s^2 + K_p s + K_i} = 0$$

⇒ Notice that Now

$$\frac{\Theta(s)}{\Theta_d(s)} = \frac{\cancel{1/s} G(s) K(s)}{1 + G(s) K(s)}$$

$$= \frac{K_d s^2 + K_p s + K_i}{s^3 + K_d s^2 + K_p s + K_i}$$

3rd order.

So now the coefficient matching business is not valid any more. However, we know ~~that~~ the guidelines i.e.,

- $K_p \uparrow \longrightarrow t_r \downarrow$
- $K_d \uparrow \longrightarrow M_p \downarrow$
- $K_i \longrightarrow e_{ss} \downarrow$